



**University Of Technology**  
**Building and Construction Eng. Dept.**  
**Final Exam. / First Attempt — 2013/2014**



Subject : Design of Steel Structures  
Branch : structural Eng.  
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Open Books and Notes  
Answer four questions

Q<sub>1</sub>: Design the member which has a length of 20ft and is carrying a tension service dead load of 50 kip and live load of 100 kip . Section  $W_{12 \times w}$  is available. The member is connected through the web only to a gusset plate which has a thickness of 1/2 in . Use A490 bolts with 3/4 in. diameter, threads are excluded from shear plane and the connection is bearing-type. The steel is A572 Gr.42.

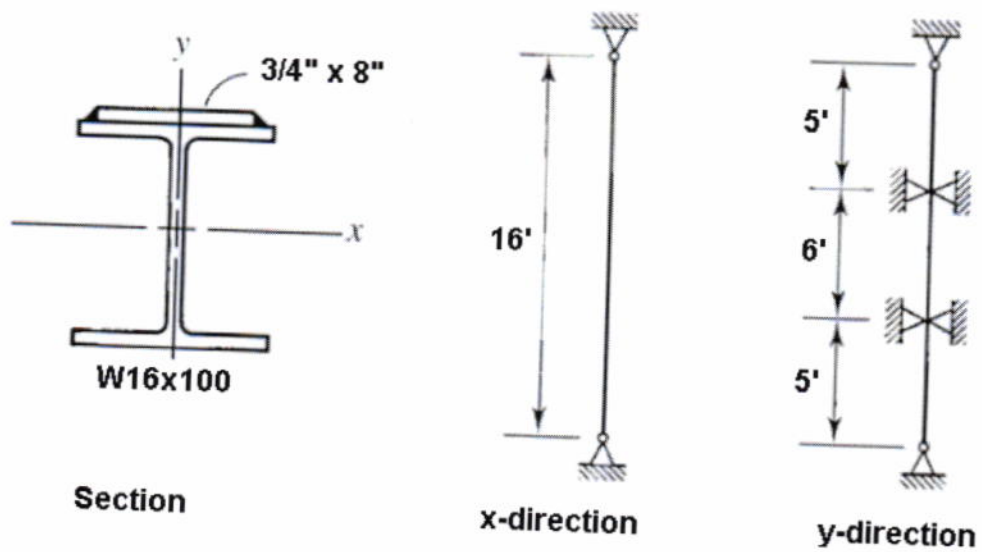
Q<sub>2</sub>: A) The column shown in Fig.1 below is fabricated by welding a 3/4-inch by 8-inch cover plate to the flange of a  $W_{16 \times 100}$  section. Steel with  $F_y = 50$  ksi is used for both components. Compute the strength of the column based on flexural buckling only. Assume that the components are connected in such a way that the member is fully effective.

B) A column base plate with  $N=25$ in. and  $B=25$ in. is resting on 30in. x30in. pedestal. The column section is  $HP_{14 \times 117}$ . The column has a total factored load of 1030k. Design the thickness of the base plate using  $F_y=50$ ksi and  $f'_c=3.5$ ksi.

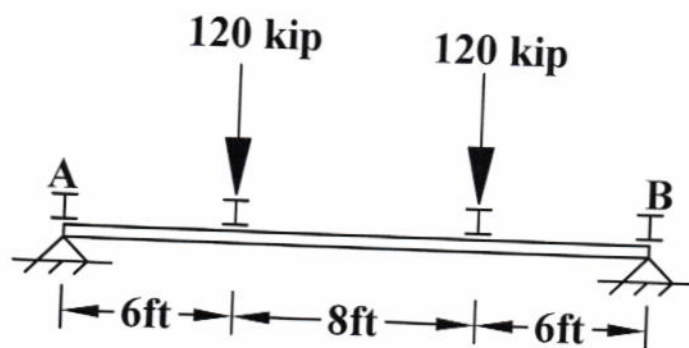
Q<sub>3</sub>: the beam AB shown in Fig.2 has steel section  $W_{18 \times 143}$  . Lateral support at each end and under the concentrated loads are provided. The service load 120kip is 20% due to dead load and 80% due to live load. Check for flexural and shear strengths, deflection and local web yielding at the support. use A992 steel and  $N=5$ in.

Q<sub>4</sub>: The simply supported member shown in Fig. 3 is subjected to the end couples (bending is about the strong axis) and the axial load. Design this member if these moments and axial loads are from service loads and consist of equal parts dead load and live load. Lateral supports are provided only at the ends. Use  $F_y=50$  ksi,  $C_b=1$  and  $W_{12 \times w}$  steel section.

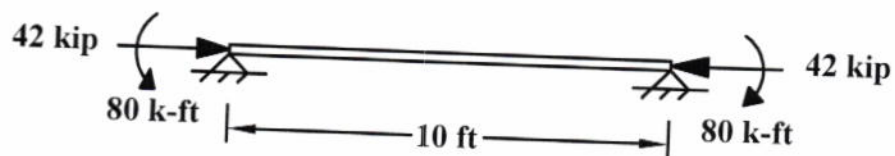
Q<sub>5</sub>: A beam-to-column connection is made with a structural tee as shown in the Fig.4 . Eight 1-inch-diameter, A490, fully tightened bearing-type bolts are used to attach the flange of the tee section to the column flange. Compute the maximum allowable factored load of this connection (the tee-to-column connection) at an eccentricity of 3.0 inches. Assume that the bolt threads are in the plane of the shear. The steel is A992.



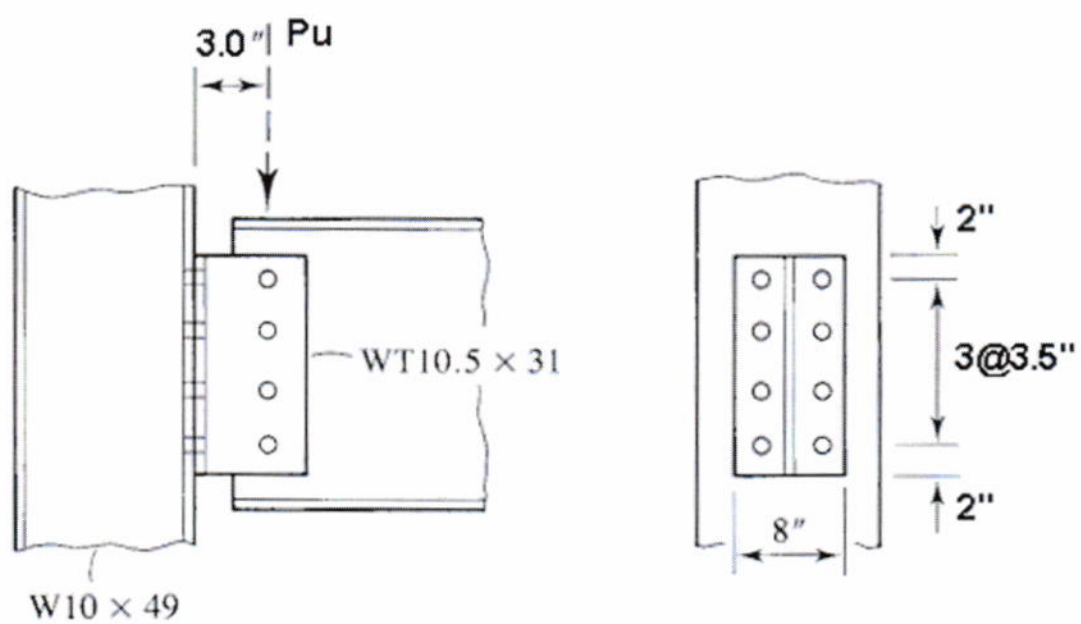
**Fig. 1**



**Fig.2**



**Fig.3**



**Fig.4**

Q<sub>1</sub>: Solution,

$$F_y = 42 \quad F_u = 60$$

$$P_u = 1.2D + 1.6L$$

$$= 1.2(50) + 1.6(100) = 220 \text{ kIP}$$

$$A_g = \frac{P_u}{0.9F_y} = \frac{220}{0.9 \times 42} = 5.82 \text{ in}^2$$

$$A_e = \frac{P_u}{0.75F_u} = \frac{220}{0.75 \times 60} = 4.88 \text{ in}^2$$

$$r_{\min} \geq L/300 \leq \frac{20 \times 12}{300} = 0.8$$

$$\text{Try } W_{12 \times 22}, \quad A_g = 6.48 \quad d = 12.3 \quad t_w = 1/4 \quad t_f = 0.425$$

$$r_x = 4.91 \quad r_y = 0.848 \quad T = 10^{3/8}$$

$$A = 6.48 > 5.82 \quad \text{o.k.}$$

$$r_y = 0.848 > 0.8 \quad \text{o.k.}$$

assume more than 4 fasteners in direction of loading,  $\therefore U = 0.7$  <sup>Table</sup>

$$A_n = A_g - t \leq d + 1/8 + t \leq \frac{S^2}{4g}$$

$$A_n = 6.48 - 0.25 \left( 0.75 + \frac{1}{8} \right) \times 2 = 6.0425$$

$$\therefore A_e = 0.7 \times A_n = 4.22 < 4.88 \quad \text{not o.k.}$$

$$\text{Try } W_{12 \times 26}, \quad A_g = 7.65 \quad d = 12.2 \quad t_w = 1/4 \quad r_y = 1.51$$

$$t_f = 0.38 \quad T = 10^{1/8}$$

$$A_e = 0.7 [7.65 - 0.25 \times 2 (0.75 \times 1.25)] = 5.04 > 4.88 \quad \text{o.k.}$$

$$* L_{e_{min}} = 1 \frac{1}{4} \text{ (Table)} \Rightarrow \text{Use } 2''$$

$$S = 3d = 3 * \frac{3}{4} = 2.25'' \Rightarrow \text{Use } 2.5''$$

\* Bolts,

$$\text{shear, } R_n = \frac{\pi (.75)^2}{4} * \overset{\text{Table}}{75} = 33.13 \text{ kip, } \phi R_n = .75 * 33.13 = 24.8 \text{ kip}$$

Bearing, Use  $t = 1/4$

$$h_d = \frac{3}{4} + \frac{1}{16} = \frac{13}{16} = 0.8125$$

$$R_n = 1.2 L_c t F_u \leq 2.4 d t F_u$$

$$R_n = 1.2 (2 - \frac{13}{32}) \frac{1}{4} * 60 = 28.68$$

$$2.4 * \frac{3}{4} * \frac{1}{4} * 60 = 27 \text{ kip}$$

$$28.68 > 27, \therefore \text{Use } 27 \text{ kip}$$

$$R_n = 1.2 (2.5 - \frac{13}{16}) * .25 * 60 = 30.37 > 27$$

$$\therefore \text{Use } 27$$

$$\therefore \phi R_n = .75 * 27 = 20.25 \text{ kip}$$

Use the lower of shear & bearing,

$$\therefore \text{No. of bolts} = \frac{220}{20.25} = 10.86$$

$$\therefore \text{Use } 12 \text{ bolts}$$

$$U = 1 - \frac{\bar{x}}{l} = 1 - \frac{0.625}{12.5} = 0.95$$

$$0.95 > 0.7 \quad \text{o.k.}$$

$$\therefore P_u = 0.75 A_e F_u$$

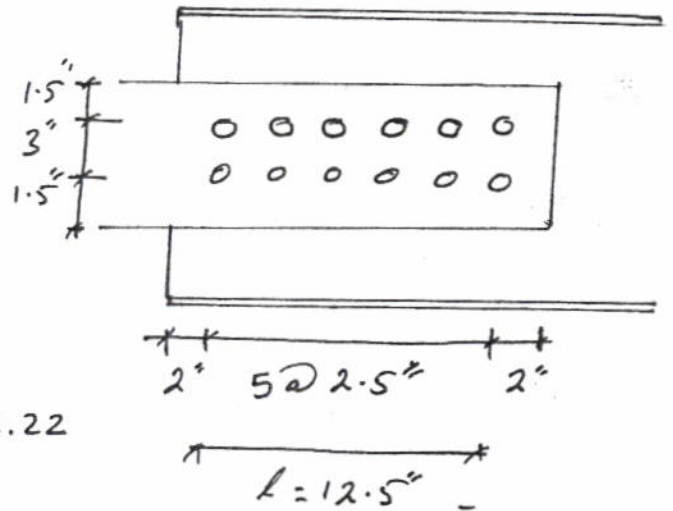
$$A_e = 0.95 [7.65 - 0.25 \times 2 (0.75 \times 12.5)] = 7.22$$

$$P_u = 0.75 \times 7.22 \times 60$$

$$= 324.9 > 220$$

$$P_u = 0.9 A_g F_y = 344.3 > 220$$

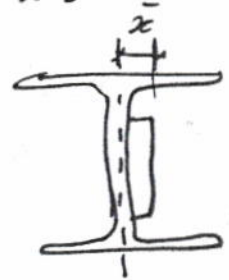
$$\therefore \text{USE } W_{12 \times 26}$$



$$\bar{x} = \frac{tw}{2} + t_{gall.}$$

$$= \frac{1}{8} + \frac{1}{2}$$

$$\bar{x} = 0.625$$



Q2:A

Shape	Area, <i>A</i>	Depth, <i>d</i>		Web		Flange				Distance					
				Thickness, <i>t<sub>w</sub></i>	$\frac{t_w}{2}$	Width, <i>b<sub>f</sub></i>	Thickness, <i>t<sub>f</sub></i>	<i>k</i>		<i>k<sub>1</sub></i>	<i>T</i>	Work- able Gage			
	<i>k<sub>des</sub></i>	<i>k<sub>dot</sub></i>													
	in. <sup>2</sup>	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	
W16×100	29.5	17.0	17	0.585	<sup>9</sup> / <sub>16</sub>	<sup>5</sup> / <sub>16</sub>	10.4	10 <sup>3</sup> / <sub>8</sub>	0.985	1	1.39	1 <sup>7</sup> / <sub>8</sub>	1 <sup>1</sup> / <sub>8</sub>	13 <sup>1</sup> / <sub>4</sub>	5 <sup>1</sup> / <sub>2</sub>

Nom- inal Wt.	Compact Section Criteria		Axis X-X				Axis Y-Y				<i>r<sub>ts</sub></i>	<i>h<sub>o</sub></i>	Torsional Properties	
			<i>I</i>	<i>S</i>	<i>r</i>	<i>Z</i>	<i>I</i>	<i>S</i>	<i>r</i>	<i>Z</i>			<i>J</i>	<i>C<sub>w</sub></i>
	lb/ft	$\frac{b_f}{2t_f}$	$\frac{h}{t_w}$	in. <sup>4</sup>	in. <sup>3</sup>	in.	in. <sup>3</sup>	in. <sup>4</sup>	in. <sup>3</sup>	in.	in. <sup>3</sup>	in.	in.	in. <sup>4</sup>
100	5.29	24.3	1490	175	7.10	198	186	35.7	2.51	54.9	2.92	16.0	7.73	11900

With the location of the horizontal centroidal axis known, the moment of inertia with respect to this axis can be found

$$I'_x = I_x + A(\text{distance})^2$$

Component	<i>A</i>	<i>y</i>	<i>Ay</i>	<i>I</i>	distance	<i>I</i> + <i>A</i> (distance) <sup>2</sup>
Plate	6	0.375	2.25	0.28125	7.375	326.625
W	29.5	9.25	272.875	1490	1.5	1556.375
sum	35.5	7.75	275.125			1883

For the vertical axis,

$$I_y = \frac{1}{12} \left( \frac{3}{4} \right) (8)^3 + 186 = 218 \text{ in.}^4$$

$$r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{1883}{35.5}} = 7.283 \text{ in.}$$

$$r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{218}{35.5}} = 2.478 \text{ in.}$$

$$\frac{K_x L_x}{r_x} = \frac{(16 \times 12)}{7.283} = 26.362$$

$$\frac{K_y L_y}{r_y} = \frac{(6 \times 12)}{2.478} = 29.055$$

$$\text{Maximum } \frac{KL}{r} = \frac{K_y L_y}{r_y} = 29.055 < 200$$

$$P_0 = F_y A_g = 50 \times 35.5 = 1775 \text{ kip}$$

$$P_e = \frac{\pi^2 E}{\left( \frac{KL}{r} \right)^2} A_g = 3.1415^2 \times \frac{29000}{29.055^2} (35.5) = 12036.2 \text{ kip}$$

$$\frac{P_e}{P_0} = \frac{12036.2}{1775} = 6.78 > 0.44 \text{ then:}$$

$$P_n = \left[ 0.658 \left( \frac{P_0}{P_e} \right) \right] P_0 = \left[ 0.658 \left( \frac{1775}{12036.2} \right) \right] (1775) = 1668.75 \text{ kip}$$

$$P_r = \phi_c P_n = 0.9 \times 1668.91 = 1501.88 \text{ kip}$$

Q2:B

Shape	Area, A	Depth, d		Web			Flange				Distance			
				Thickness, t <sub>w</sub>		t <sub>w</sub> 2	Width, b <sub>f</sub>		Thickness, t <sub>f</sub>		k	k <sub>1</sub>	T	Workable Gage
	in. <sup>2</sup>	in.	in.	in.	in.		in.	in.	in.	in.	in.	in.	in.	
HP14×117 <sup>f</sup>	34.4	14.2	14 1/4	0.805	13/16	7/16	14.9	14 7/8	0.805	13/16	1 1/2	1 1/16	11 1/4	5 1/2

Solution:

$$N = 25 \text{ in.}$$

$$\text{Verify } N \geq d + 2(3.00 \text{ in.}):$$

$$25 \text{ in.} \stackrel{?}{\geq} 14.2 + 2(3 \text{ in.}) = 20.2 \text{ in.} \quad \text{ok}$$

$$B = 25 \text{ in.}$$

$$\text{Verify } B \geq b_f + 2(3.00 \text{ in.}):$$

$$25 \text{ in.} \stackrel{?}{\geq} 14.9 + 2(3 \text{ in.}) = 20.9 \text{ in.} \quad \text{ok}$$

$$P_p = 0.85 f'_c A_1 \sqrt{\frac{A_2}{A_1}}$$

$$P_p = 0.85 \times 3.5 \times 25 \times 25 \sqrt{\frac{30 \times 30}{25 \times 25}} = 2231.25 \text{ kip} \quad \frac{A_2}{A_1} = \frac{30 \times 30}{25 \times 25} = 1.44 \quad \text{ok}$$

$$\phi_c P_p \geq P_u$$

$$0.6 \times 2231.25 \text{ kip} \geq 1030 \text{ kip}$$

$$1338.75 \text{ kip} > 1030 \text{ kip} \quad \text{ok}$$

$$m = \frac{N - 0.95d}{2} = \frac{25 - 0.95 \times 14.2}{2} = 5.755 \text{ in.}$$

$$n = \frac{B - 0.8b_f}{2} = \frac{25 - 0.8 \times 14.9}{2} = 6.54 \text{ in.}$$

$$n' = \frac{\sqrt{db_f}}{4} = \frac{\sqrt{14.2 \times 14.9}}{4} = 3.636 \text{ in.}$$

$$x = \left( \frac{4db_f}{(d + b_f)^2} \right) \frac{P_u}{\phi_c P_p} \leq 1.0$$

$$x = \left( \frac{4 \times 14.2 \times 14.9}{(14.2 + 14.9)^2} \right) \frac{1030}{1338.75} \leq 1.0$$

$$x = 0.769 < 1 \quad \text{use } x = 0.769$$

$$\lambda = \frac{2\sqrt{x}}{1 + \sqrt{1 - x}} \leq 1.0$$

$$\lambda = \frac{2\sqrt{0.769}}{1 + \sqrt{1 - 0.769}} \leq 1.0$$

$$\lambda = 1.184 \leq 1.0 \quad \text{use } \lambda = 1.0$$

$$\lambda n' = 1.0 \times 3.636 = 3.636 \text{ in.}$$

$$t_{\min} = l \sqrt{\frac{2P_u}{0.9F_yBN}} = 6.54 \times \sqrt{\frac{2 \times 1030}{0.9 \times 50 \times 25 \times 25}} = 1.77 \text{ in.}$$

Use  $t = 2 \text{ in.}$

Use plate 25 in.  $\times$  25 in.  $\times$  2 in.

Q3: Solution,

$$F_y = 50 \quad F_u = 65 \text{ ksi}$$

$$P_u = 1.2(0.2 \times 120) + 1.6(0.8 \times 120) = 1.2 \times 24 + 1.6 \times 96$$

$$P_u = 182.4 \text{ kip}$$

$$M_u = P_u \times 6 = 1094.4 \text{ k-ft}$$

$$L_b = 8'$$

$$W18 \times 143, \quad A = 42.1 \quad d = 19.5 \quad t_w = 0.73 \quad k = 1.72 \quad \frac{b_f}{2t_f} = 4.25$$

$$h/t_w = 22 \quad I_x = 2750 \quad S_x = 282 \quad Z_x = 322$$

$$r_y = 2.72$$

$$* L_p = 1.76 \times 2.72 \sqrt{\frac{29000}{50}} = 9.6'$$

$L_b < L_p \quad \therefore$  laterally supported beam

$$\frac{b_f}{2t_f} \leq 0.38 \sqrt{\frac{29000}{50}}, \quad h/t_w \leq 3.76 \sqrt{\frac{29000}{50}}$$

$$4.25 < 9.15$$

$$22 < 90.5$$

$\therefore$  Compact section.

$$M_n = Z_x F_y = \frac{322 \times 50}{12} = 1341.66 \text{ kip-ft}$$

$$\phi M_n = 0.9 \times 1341.66 = 1207.5 > M_u = 1094.4 \quad \text{O.K.}$$

$$* \quad h/t_w \leq 2.24 \sqrt{\frac{29000}{50}}$$

$$22 < 53.93$$

$$\therefore C_D = 1 \text{ \& } \phi_D = 1$$

$$V_n = 0.6 F_y A_w C_D$$

$$= 0.6 * 50 (19.5 * 0.73) * 1$$

$$= 427.05$$

$$\phi V_n = 427 > V_u = 182.4 \quad \text{o.k.}$$

$$* \quad R_n = (2.5k + N) F_y t_w$$

$$= (2.5 * 1.72 + 5) 50 * 0.73 = 339.45 \text{ kip}$$

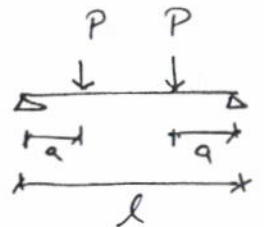
$$\phi R_n = 339.45 > 182.4 \quad \text{o.k.}$$

$$* \quad L/360 = \frac{20 * 12}{360} = 0.66 \text{ in}$$

$$P = 96 \text{ kip} \quad l = 20 * 12 = 240'' \quad a = 6 * 12 = 72''$$

$$I_x = 2750$$

$$\Delta_{\text{max center}} = \frac{Pa}{24EI} (3l^2 - 4a^2) = 0.549'' < 0.66'' \quad \text{o.k.}$$



Q4 : Solution,

$$P_u = 1.2(21) + 1.6(21) = 58.8 \text{ kip}$$

$$M_{nt} = 1.2(40) + 1.6(40) = 112 \text{ kip-ft}$$

$$l_x = l_y = 10'$$

$$k_x = k_y = 1$$

$$(kl)_y = 10$$

$$(kl)_y = \frac{(kl)_x}{r_x/r_y} = \frac{10}{1.8} = 5.5$$

$$\therefore KL = 10$$

$$\text{assume } \phi = 1.18$$

$$\phi P_u = \frac{1.18}{1000} \times 58.8 = 0.069 < 0.2, \text{ H1-1 b applied}$$

$$\frac{1}{2} \phi P_u + \frac{9}{8} (b_x M_{ux}) \leq 1$$

$$\frac{1}{2} \times \frac{1.18}{1000} \times 58.8 + \frac{9}{8} \left( \frac{b_x}{1000} \times 112 \right) = 1$$

$$b_x = 7.66$$

With  $KL = 10$  and from Table 6-1 choose,

$$W12 \times 26, \phi = 4.64, b_x = 7.78$$

$$\phi P_u = \frac{4.64}{1000} \times 58.8 = 0.27 > 0.2, \text{ use H1-1 a}$$

$$\frac{4.64}{1000} \times 58.8 + \frac{7.78}{1000} \times 112 < 1$$

$$1.14 > 1.0 \text{ not o.k.}$$

Try larger section

•  $W_{12 \times 30}$ ,  $P = 3.99$   $b_x = 6.62$

•  $\frac{3.99}{1000} \times 58.8 = 0.23 > 0.2$ ,  $H_{1-1a}$

$$0.23 + \frac{6.62}{1000} \times 112 < 1$$

$$0.97 < 1 \quad \text{o.k.}$$

$W_{12 \times 30}$ ,  $A = 8.79$   $b_f/2t_f = 7.41$   $h/t_w = 41.8$

$$b_f/2t_f < 0.56 \sqrt{\frac{29000}{50}} \quad , \quad h/t_w < 1.49 \sqrt{\frac{E}{F_y}}$$

$$7.41 < 13.48 \quad \checkmark$$

$$41.8 > 35.87 \quad \text{not o.k.}$$

Try  $W_{12 \times 40}$ ,  $A = 11.7$   $r_x = 5.13$   $r_y = 1.94$

$$\frac{b_f}{2t_f} = 7.77 \quad h/t_w = 33.6 \quad Z_x = 57 \quad S_x = 51.1$$

$$r_{ts} = 2.21 \quad h_o = 11.4 \quad J = 0.906$$

\*  $7.77 < 13.48 \quad \checkmark$

$33.6 < 35.87 \quad \checkmark \quad \text{o.k.}$

$$(kl/r)_x = \frac{10 \times 12}{5.13} = 23.4$$

$$(kl/r)_y = \frac{10 \times 12}{1.94} = 61.8 < 300 \quad \text{o.k.}$$

$$kl/r \leq 4.71 \sqrt{\frac{E}{F_y}}$$

$$61.8 < 113.4$$

$$\therefore F_{cr} = [0.658^{F_y/F_c}] F_y$$

$$F_c = \frac{\pi^2 29000}{(61.8)^2} = 74.9$$

$$F_{cr} = [0.658^{50/74.9}] 50 = 37.8 \text{ ksi}$$

$$P_n = F_{cr} A = 442.26$$

$$\phi_c P_n = 398 \text{ kip}, \quad \frac{P_u}{\phi P_n} = \frac{58}{398} = 0.147 < 0.2 \therefore \text{use } H_{1-1b}$$

$$* L_b =$$

$$L_p = 1.76 * 1.94 \sqrt{\frac{29000}{50}} = 6.85'$$

$$L_r = 1.95 * 2.21 * \frac{29000}{35} \sqrt{\frac{0.906}{51.1 * 11.4}} \sqrt{1 + \sqrt{1 + 6.76 \left( \frac{35 * 51.1 * 11.4}{29000 * 0.906} \right)}}$$

$$L_r = 21.9'$$

$$L_r > L_b > L_p$$

$$M_p = 50 * 57 = 2850 \text{ ksi} = 237.5 \text{ k-ft}$$

$$\therefore M_u = 1 \left[ 237.5 - \left( 237.5 - \frac{35 * 51.1}{12} \right) \left( \frac{10 - 6.85}{21.9 - 6.85} \right) \right] \leq M_p$$

$$M_u = 219 < M_p$$

$$\phi M_u = 0.9 * 219 = 197.1 \text{ k-ft}$$

$$B_1 = \frac{C_m}{1 - \frac{P_u}{P_{e1}}} \geq 1$$

$$C_m = 0.6 - 0.4(-1) = 1$$

$$B_1 = \frac{1}{1 - \frac{58.8}{\frac{\pi^2 E A}{\left( \frac{k_1 \ell}{r} \right)^2}}} = \frac{1}{1 - \frac{58.8}{\frac{\pi^2 29000 * 11.9}{(23.4)^2}}} = 1.007$$

$$M_u = M_{nt} * B_1 = 113$$

$$\frac{P_u}{2 \phi P_n} + \frac{M_{ux}}{\phi_b M_{nx}} \leq 1$$

$$\frac{58.8}{2 * 398} + \frac{113}{197.1} < 1$$

$$0.646 < 1 \quad \text{o.k.}$$

$\therefore$  Use  $W12 \times 40$

Q5:

Shape	Area, <i>A</i>	Depth, <i>d</i>		Stem				Flange				Distance		
				Thickness, <i>t<sub>w</sub></i>		$\frac{t_w}{2}$	Area	Width, <i>b<sub>f</sub></i>		Thickness, <i>t<sub>f</sub></i>		<i>k</i>		Work- able Gage
	in. <sup>2</sup>	in.	in.	in.	in. <sup>2</sup>			in.	in.	in.	in.	in.	in.	
WT10.5×31 <sup>c</sup>	9.13	10.5	10½	0.400	3/8	3/16	4.20	8.24	8¼	0.615	5/8	1.12	15/16	5½

Shape	Area, A	Depth, d		Web		Flange				Distance					
				Thickness, $t_w$		$\frac{t_w}{2}$	Width, $b_f$		Thickness, $t_f$		k		$k_1$	T	Work- able Gage
								$k_{des}$	$k_{det}$						
	in. <sup>2</sup>	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.		
W10×49	14.4	10.0	10	0.340	5/16	3/16	10.0	10	0.560	9/16	1.06	1 1/4	13/16	▼	5 1/2

Determine the bearing and shear nominal strengths. For the strength in bearing, use a hole diameter of

$$h = d + \frac{1}{16} = 1 + \frac{1}{16} = 1.0625 \text{ in.}$$

$$L_e = 2 \text{ in.}$$

$$L_c = L_e - \frac{h}{2} = 2 - \frac{1.0625}{2} = 1.46875 \text{ in.}$$

$$R_n = 1.2L_c t F_u = 1.2 \times 1.46875 \times 0.56 \times 65 = 64.155 \text{ kip}$$

$$\text{Upper limit} = 2.4dt F_u = 2.4 \times 1 \times 0.56 \times 65 = 87.36 \text{ kip}$$

$$\therefore \text{for this bolt } R_n = 64.155 \text{ kip}$$

For the other holes, use  $s = 3.5 \text{ in.}$  Then,

$$L_c = S - h = 3.5 - 1.0625 = 2.4375$$

$$R_n = 1.2L_c t F_u = 1.2 \times 2.4375 \times 0.56 \times 65 = 106.47 \text{ kip}$$

$$\text{Upper limit} = 2.4dt F_u = 2.4 \times 1 \times 0.56 \times 65 = 87.36 \text{ kip}$$

$$\therefore \text{for this bolt } R_n = 87.36 \text{ kip}$$

**TABLE J3.2**  
**Nominal Stress of Fasteners and Threaded Parts,**  
**ksi (MPa)**

Description of Fasteners	Nominal Tensile Stress, $F_{nt}$ , ksi (MPa)	Nominal Shear Stress in Bearing-Type Connections, $F_{nv}$ , ksi (MPa)
A307 bolts	45 (310) [a][b]	24 (165) [b][c][f]
A325 or A325M bolts, when threads are not excluded from shear planes	90 (620) [e]	48 (330) [f]
A325 or A325M bolts, when threads are excluded from shear planes	90 (620) [e]	60 (414) [f]
A490 or A490M bolts, when threads are not excluded from shear planes	113 (780) [e]	60 (414) [f]
A490 or A490M bolts, when threads are excluded from shear planes	113 (780) [e]	75 (520) [f]
Threaded parts meeting the requirements of Section A3.4, when threads are not excluded from shear planes	$0.75 F_u$ [a][d]	$0.40 F_u$
Threaded parts meeting the requirements of Section A3.4, when threads are excluded from shear planes	$0.75 F_u$ [a][d]	$0.50 F_u$

[a] Subject to the requirements of Appendix 3.

[b] For A307 bolts the tabulated values shall be reduced by 1 percent for each  $1/16$  in. (2 mm) over 5 diameters of length in the grip.

[c] Threads permitted in shear planes.

[d] The nominal tensile strength of the threaded portion of an upset rod, based upon the cross-sectional area at its major thread diameter,  $A_D$ , which shall be larger than the nominal body area of the rod before upsetting times  $F_y$ .

[e] For A325 or A325M and A490 or A490M bolts subject to tensile fatigue loading, see Appendix 3.

[f] When bearing-type connections used to splice tension members have a fastener pattern whose length, measured parallel to the line of force, exceeds 50 in. (1270 mm), tabulated values shall be reduced by 20 percent.

For the strength in shear,:

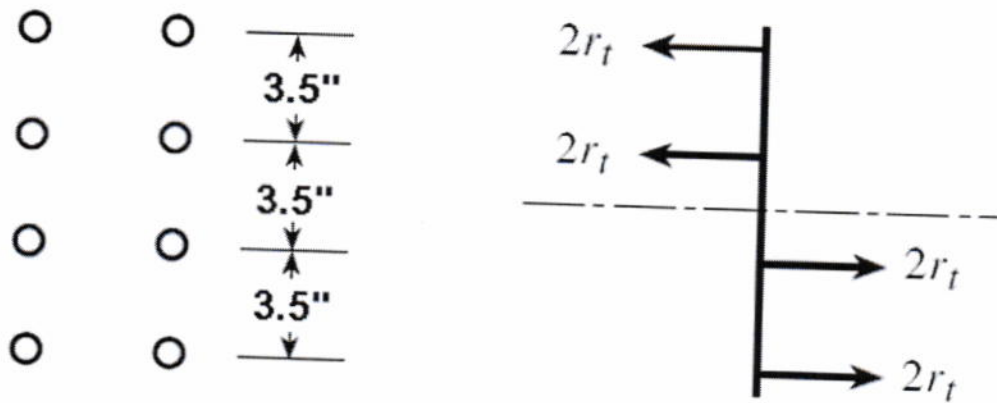
$$R_n = F_{nv} A_b = 60 \times \left( \frac{\pi \times 1}{4} \right) = 47.12 \text{ kip}$$

$$R_{n_{bearing_{total}}} = 2 \times 64.155 + 6 \times 87.36 = 652.47 \text{ kip}$$

$$R_{n_{shear}total} = 8 \times 47.12 = 376.96 \text{ kip}$$

$$R_{rtotal} = 0.75 \times 376.96 = 282.743 \text{ kip}$$

Compute the tensile force per bolt and then check the tension-shear interaction. Because of symmetry, the centroid of the connection is at middepth. Figure below shows the bolt areas and the distribution of bolt tensile forces.



The moment of the resisting couple is found by summing moments about the neutral axis:

$$\sum M_{NA} = 2r_t(1.75 + 5.25 + 5.25 + 1.75) = 28r_t$$

The applied moment is:

$$M_u = P_u e = 3P_u$$

Equating the resisting and applied moments, we get:

$$3P_u = 28r_t$$

The factored load shear stress is:

$$f_v = \frac{P_u}{N \times A_b} = \frac{P_u}{8 \times 0.785}$$

$$F'_{nt} = 1.3F_{nt} - \frac{F_{nt}}{\phi F_{nv}} f_v \leq F_{nt}$$

$$F'_{nt} = 1.3 \times 113 - \frac{113}{0.75 \times 60} \frac{P_u}{8 \times 0.785} \leq F_{nt}$$

$$146.9 - 0.316P_u = 113$$

$$P_u = 107.28 \text{ kip}$$

The design tensile strength is:

$$R_r = \phi F'_{nt} A_b = 0.75 \left( 1.3 \times 113 - \frac{113}{0.75 \times 60} \frac{P_u}{8 \times 0.785} \right) \times 0.785 \geq r_t$$

$$0.75 \left( 1.3 \times 113 - \frac{113}{0.75 \times 60} \frac{P_u}{8 \times 0.785} \right) \times 0.785 \geq \frac{3P_u}{28}$$

$$\left( 1.3 \times 113 - \frac{113}{0.75 \times 60} \frac{P_u}{8 \times 0.785} \right) = \frac{P_u}{5.495}$$

$$146.9 = 0.497P_u$$

$$P_u = 295.573 \text{ kip}$$

$$P_u = 107.28 \text{ kip}$$